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UTILITY PATENT APPLICATION TRANSMITTAL (Small Entity)

(Only for new nonprovisional applications under 37 CFR 1.53(b))

Docket No.
240,988218Total Pages in this Submission
3**TO THE ASSISTANT COMMISSIONER FOR PATENTS**Box Patent Application
Washington, D.C. 20231

Transmitter herewith for filing under 35 U.S.C. 111(a) and 37 C.F.R. 1.53(b) is a new utility patent application for an invention entitled:

BALANCER FOR ORBITAL ABRADING MACHINE

and invented by:

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 09/236947
 01/25/99
If a **CONTINUATION APPLICATION**, check appropriate box and supply the requisite information:
☒ Continuation ☐ Divisional ☐ Continuation-in-part (CIP) of prior application No.: _____

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Enclosed are:

Application Elements

1. ☒ Filing fee as calculated and transmitted as described below
2. ☒ Specification having 21 pages and including the following:
 - a. ☒ Descriptive Title of the Invention
 - b. ☐ Cross References to Related Applications (if applicable)
 - c. ☐ Statement Regarding Federally-sponsored Research/Development (if applicable)
 - d. ☐ Reference to Microfiche Appendix (if applicable)
 - e. ☒ Background of the Invention
 - f. ☒ Brief Summary of the Invention
 - g. ☒ Brief Description of the Drawings (if drawings filed)
 - h. ☒ Detailed Description
 - i. ☒ Claim(s) as Classified Below
 - j. ☒ Abstract of the Disclosure

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Application Elements (Continued)

3. ☒ Drawing(s) *(when necessary as prescribed by 35 USC 113)*

a. ☐ Formal b. ☒ Informal Number of Sheets 4

4. ☒ Oath or Declaration

a. ☒ Newly executed *(original or copy)* ☐ Unexecuted

b. ☐ Copy from a prior application (37 CFR 1.63(d)) *(for continuation/divisional application only)*

c. ☒ With Power of Attorney ☐ Without Power of Attorney

d. ☐ DELETION OF INVENTOR(S)

Signed statement attached deleting inventor(s) named in the prior application,
see 37 C.F.R. 1.63(d)(2) and 1.33(b).

5. ☐ Incorporation By Reference *(usable if Box 4b is checked)*

The entire disclosure of the prior application, from which a copy of the oath or declaration is supplied under Box 4b, is considered as being part of the disclosure of the accompanying application and is hereby incorporated by reference therein.

6. ☐ Computer Program in Microfiche

7. ☐ Genetic Sequence Submission *(if applicable, all must be included)*

a. ☐ Paper Copy

b. ☐ Computer Readable Copy

c. ☐ Statement Verifying Identical Paper and Computer Readable Copy

Accompanying Application Parts

8. ☒ Assignment Papers *(cover sheet & documents)*

9. ☐ 37 CFR 3.73(b) Statement *(when there is an assignee)*

10. ☐ English Translation Document *(if applicable)*

11. ☒ Information Disclosure Statement/PTO-1449 ☒ Copies of IDS Citations

12. ☐ Preliminary Amendment

13. ☒ Acknowledgment postcard

14. ☒ Certificate of Mailing

☐ First Class ☒ Express Mail *(Specify Label No.):* EL249542682US

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Accompanying Application Parts (Continued)

15. ☐ Certified Copy of Priority Document(s) (if foreign priority is claimed)
16. ☒ Small Entity Statement(s) - Specify Number of Statements Submitted: 1
17. ☐ Additional Enclosures (please identify below):

Fee Calculation and Transmittal

CLAIMS AS FILED

For	#Filed	#Allowed	#Extra	Rate	Fee
Total Claims	16	- 20 =	0	x \$9.00	\$0.00
Indep. Claims	7	- 3 =	4	x \$39.00	\$156.00
Multiple Dependent Claims (check if applicable) <input type="checkbox"/>					\$0.00
BASIC FEE					\$380.00
OTHER FEE (specify purpose) <u>Assignment Recordal Form</u>					\$40.00
TOTAL FILING FEE					\$576.00

- ☒ A check in the amount of \$576.00 to cover the filing fee is enclosed.
- ☒ The Commissioner is hereby authorized to charge and credit Deposit Account No. 02-1550 as described below. A duplicate copy of this sheet is enclosed.
- ☐ Charge the amount of _____ as filing fee.
- ☒ Credit any overpayment.
- ☒ Charge any additional filing fees required under 37 C.F.R. 1.16 and 1.17.
- ☐ Charge the issue fee set in 37 C.F.R. 1.18 at the mailing of the Notice of Allowance, pursuant to 37 C.F.R. 1.311(b).

Dated: January 25, 1999


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BALANCER FOR ORBITAL ABRADING MACHINEBACKGROUND OF THE INVENTION

Orbital abrading machines are well-known and generally comprise a portable, manually manipulatable housing, a motor supported by the housing and having or being coupled to a drive shaft driven for rotation about a first axis, and an assembly for mounting a pad for abrading a work surface for orbital movement about the first axis. In a random orbital abrading machine, the assembly serves to additionally mount the pad for free rotational movement about a second axis, which is disposed parallel to the first axis.

The assembly typically includes a head portion coupled for driven rotation with the drive shaft about the first axis and defining a mounting recess having an axis arranged coincident with the second axis, a bearing supported within the mounting recess, and means for connecting the pad to the bearing for rotation about the second axis.

Orbital machines by nature are subject to dynamic unbalance and require the inclusion of a counterbalance system to reduce vibration to an acceptance level. The typical design approach has been to account only for the unbalance, which is created by the mass centers of the pad and portions of the assembly not disposed concentric to the first axis, by the addition of balancing masses to the housing. This approach can create a machine that is balanced, that is, has acceptably low vibration levels,

while the machine is running at free speed in an unloaded condition. However, once the machine is loaded, as a result of placing the pad in abrading engagement with a work surface, additional forces are introduced and the machine becomes unbalanced and this unbalance is detected by the operator in the form of vibration. This is undesirable and in severe cases, may lead to vibration induced injuries such as carpal tunnel syndrome and white finger.

The counterbalance system referred to above, which may be used in the design of both orbital and random orbital machines, is described for example in Chapter 12 of Mechanisms and Dynamics of Machinery, Third Edition, by Hamilton H. Mabie and Fred W. Ocvirk, published by John Wiley & Sons.

Another approach is that adopted for the Atlas Copco Turbo Grinder GTG40, which uses an SKF Nova AB auto-balancing unit to reduce vibration under various loading conditions. This unit features the use of a plurality of ball bearings, which are arranged within an annular raceway and free to move therewithin as required to reducing vibrations.

SUMMARY OF THE INVENTION

It is known that both orbital and random orbital abrading machines, which include for example, sanding, grinding and buffing machines, that have been balanced to minimize vibration under no load operating conditions, may

be subjected to unacceptable levels of vibration under actual working conditions.

The present invention relates to an improved, orbital abrading machine, and more particularly to an improved random orbital buffer, which may be counterbalanced in such a manner as to minimize vibrations under actual working conditions.

The present invention is based on the realization that known balancing techniques, which may be employed to achieve proper balancing under unloaded conditions, do not take into consideration forces at work, during actual working conditions, which oftentimes result in a properly balanced machine becoming unbalanced to an unacceptable degree during use. More particularly, the present invention is directed towards a counterbalancing system adapt to minimize vibration of a orbital abrading machine under determined operating conditions.

BRIEF DESCRIPTION OF THE DRAWINGS

The nature and mode of operation of the present invention will now be more fully described in the following detailed description taken with the accompanying drawings wherein:

Fig. 1 is an exploded prospective view of a random orbital abrading machine embodying the present invention;

Fig. 2 is a balance sketch illustrating a known mode of counterbalancing an orbital abrading machine having two mass centers arranged in an offset relationship relative to an axis of rotation or first axis;

Fig. 3 is a balance sketch illustrating the present mode of counterbalancing an orbital abrading machine having mass centers arranged in the same manner as that shown in Fig. 2;

Fig. 4a is an end view of a head portion of an assembly employed to couple an abrasive pad to a drive motor of an orbital abrading machine, which is provided with a pair of masses arranged in accordance with a known counterbalancing system;

Fig. 4b is a sectional view taken along the line A-A in Fig. 4a;

Fig. 5a is an end view of a head portion of an assembly employed to couple an abrasive pad to a drive motor of an orbital abrading machine, which is provided with a pair of masses arranged in accordance with the present invention to minimize vibration of the machine under intended working conditions; and

Fig. 5b is a sectional view taken along the line A-A in Fig. 5a.

DETAILED DESCRIPTION

Reference is first made to Fig. 1, wherein an orbital abrading machine is generally designated as 10 and shown as generally including a manually manipulated housing 12, a motor 14 mounted within the housing and including or being suitably coupled to a drive shaft 16 driven for rotation about a first axis 18, and an assembly 20 which serves to connect an abrasive pad 22 to drive shaft 16 such that the pad is caused to orbit about the first axis.

Preferably machine 10 is in the form of a random orbital machine in which abrasive pad 22 is supported by assembly 20 for free rotational movement about a second axis 24, which is disposed parallel to and orbits about first axis 18. Housing 12 may be fitted with a manually manipulatable ~~hurdle~~ ^{handle} 26 and motor may be a pneumatically driven motor connected to a suitable supply of air under pressure.

Assembly 20 may be similar to that described in commonly assigned U.S. Patent 4,854,085 in that it generally includes a head portion 30 mechanically coupled to or formed integrally with drive shaft 16 and formed with a generally cylindrical mounting recess, which is designated as 32 only in Figs. 4b and 5b. This mounting recess has an axis disposed coincident with second axis 24 and is sized to mount a bearing 34 therewithin. Bearing 34 serves in turn to support means for connecting pad 22 to bearing 34, such as may be defined by a mounting shaft 36, which is disposed for rotation concentrically of axis 24 and formed with an axially extending threaded mounting opening, not shown, for removably receiving an abrasive pad mounting fastener 38. Also shown in Fig. 1 are known seal and seal mounting devices 40 for use in preventing the ingress of undesired materials upwardly into bearing 34 and an annular shroud 42 adapted to be mounted on housing 12 to extend peripherally of pad 22.

A machine having an element, such as pad 22, driven for movement about an orbital path of travel is by nature unbalanced and tends to produce vibrations, which may be felt by the hands of an operator of the machine. With a view towards maintaining such vibrations at acceptable

levels, it has been common practice to employ a counterbalance system of the type described in Chapter 12 of Mechanisms and Dynamics of Machinery, Third Edition, by Hamilton H. Mabie and Fred W. Ocvirk, published by John Wiley and Sons, which is incorporated by reference herein.

To facilitate understanding of this prior system and its use in counterbalancing of a sample orbital machine, reference is made to the balance sketch illustrated in Fig. 2 and TABULATION I set forth below:

TABULATION I General Random Orbital

Input

mass 1			mass 2			Balancing plane			Z
m1 (g)	202		m2 (g)	75.8		A (mm)			32
r1 (mm)	7		r2 (mm)	7		B (mm)			44.8
θ1 (°)	0		θ2 (°)	0		C = B-A (mm)			12.8
Z1 (mm)	19.4		Z2 (mm)	43					

Balancing Table

Plane	m (g)	r (mm)	mr (g*mm)	Z (mm)	θ	Balancing Plane A			Balancing Plane B		
						b	mr _b	(mr _b)cosθ	(mr _b)sinθ	a	mr _a
From Input											
1	202	7	1414	19.4	0	25.40	35915.60	0.00	-12.60	-17816.40	0.00
2	75.6	7	529.2	43	0	1.80	952.56	0.00	11.00	5821.20	0.00
Summation (Σ)							36868.16	0.00		-11895.20	0.00
Calculated Values											
Balancer A			2880.3 ^{mm}	32	180.0 [°]	12.80	36868.16	0.00	0.00	0.00	0.00
Balancer B			937.1 ^{mm}	44.8	0.0 [°]	0.00	0.00	0.00	12.80	11895.20	0.00
SUM							0.00	0.00		0.00	0.00

Solution Summary

Plane	mr (g*mm)	θ (°)
Balancer A	2880.3	180.00
Balancer B	937.1	0.00

where

$$\begin{aligned}
 \star (mr)A &= (((\Sigma mr b \cos \theta)^2 + (\Sigma mr b \sin \theta)^2)^{.5})/C \\
 \star\star (mr)B &= (((\Sigma mr a \cos \theta)^2 + (\Sigma mr a \sin \theta)^2)^{.5})/C \\
 \bullet \tan(\theta)A &= -(\Sigma mr a \sin \theta) / -(\Sigma mr a \cos \theta) \\
 \bullet\bullet \tan(\theta)B &= -(\Sigma mr b \sin \theta) / -(\Sigma mr b \cos \theta)
 \end{aligned}$$

portions of the masses not disposed concentrically of axis 18

8

It will be understood that m_1 is a first mass defined by pad 22, bearing 34, mounting shaft 36, mounting fastener 38, and seal and seal mounting devices 40; m_2 is a second mass defined by housing 30; r_1 and r_2 are the radial distances of the centers of masses m_1 and m_2 from the first rotational axis 18; and z_1 and z_2 are the distances of transverse planes in which masses m_1 and m_2 are disposed from a selected parallel reference plane disposed normal to axis 18, such as may be conveniently defined by a working surface of pad 22 to be presented for abrading engagement with a work surface, not shown. For the case of the sample orbital machine, the center of the pad working surface is located at point 50 shown in Fig.2, and the centers of masses m_1 and m_2 are assumed to lie in approximate alignment with second axis 24, such that the angle θ for each mass can be assumed to be essentially 0° .

The sample orbital machine may be balanced by adding two or more balancing masses, as for instance m_A and m_B , whose centers lie at suitable radial distances r_A and r_B from first axis 18 and within selected planes disposed parallel and spaced through distances z_A and z_B from the above reference plane. The number of balancing masses and their relative positions may be varied depending on installation requirements and choice of the designer of the machine. The requirement for obtaining a balanced machine is that masses m_A and m_B be sized and arranged such that the sum of the values of the columns $(mrb) \cos \theta$, $(mrb) \sin \theta$, $(mra) \cos \theta$ and $(mra) \sin \theta$ for m_1 , m_2 and m_A and m_B appearing in the Balancing Table of TABULATION I be equal to zero. As the values of these columns progressively

increase from zero, vibration caused by unbalance progressively increases.

In the solution shown in the Solution Summary of TABULATION I and illustrated in Fig. 4a, the centers of masses m_A and m_B are arranged at 180° and 0° degrees relative to axis 18, and these masses are symmetrical relative to a plane 60 in which parallel axes 18 and 24 are disposed.

An orbital or random orbital machine once balanced in accordance with the above-referenced prior practice, will remain in balance regardless of the rotation speed of the drive shaft, so long as pad 22 is permitted to rotate under unloaded conditions. However, as soon as pad 22 is loaded, as by being placed in abrading engagement with a work surface, the original balance is lost and an operator is exposed to varying degrees of vibration depending on the working conditions under which the orbital machine is used.

With certain orbital machines, such as sanders, the degree of unbalance, and thus vibration experienced by an operator under typical working conditions, is normally found to be within acceptable limits. However, for other orbital machines, such as for example, buffers, the degree of unbalance is typically found to be greater and may reach a level at which prolonged use of the machine may cause serious vibration induced injury to an operator.

The present invention seeks to provide an orbital or random orbital machine, which is adapted to be balanced while exposed to predetermined working conditions under which the machine is intended for use, so as to minimize

vibrations to which an operator is exposed, while actually using the machine for performing a given type of abrading operation.

In attempting to solve the problem of an unacceptably high vibration level experienced with the use of a random orbital buffer intended for use in the finishing of painted vehicle surfaces, it was realized that the above-described prior balancing technique for orbital machines did not take into account working loads, such as drag caused by bearing engagement of the abrading or buffing pad with the painted surface, and that it was necessary to consider the angular velocity of masses m_1 , m_2 , m_A and m_B in order to determine the values and positions required to be assumed by balancing masses m_A and m_B in order to achieve balance under actual working conditions..

To facilitate understanding of the present invention, reference is made to the balance sketch of Fig. 3 and TABULATIONS II and III set forth below:

TABULATION II

Orbital with Drag Force

Input

mass 1		mass 2		Balancing plane		Z	Loading	
m1 (g)	202.0	m2 (g)	75.6	A (mm)	32.0	RPM under load	5,000	
r1 (mm)	7.0	r2 (mm)	7.0	B (mm)	44.8	Drag force (N)	63.0	
θ1 (°)	0.0	θ2 (°)	0.0	C = B-A (mm)	12.8	angle (°)	90.0	
Z1 (mm)	19.4	Z2 (mm)	43.0			Placement (mm)	0.0	

Balancing Table

Plane	m (g)	r (mm)	mr (g*mm)	ω ² (rad/s) ²	Force (N)	Z (mm)	θ	Balancing Plane A				Balancing Plane B			
					Force (N)			b	force*b	(force*b)cosθ	(force*b)sinθ	a	force*a	(force*a)cosθ	(force*a)sinθ
From Input															
1	202	7	1,414.0	274,156	387.7	19.4	0.0	25.40	9,846.5	9,846.5	0.0	-12.60	-4,884.5	-4,884.5	0.0
2	75.6	7	529.2	274,156	145.1	43.0	0.0	1.80	261.1	261.1	0.0	11.00	1,595.9	1,595.9	0.0
Drag					63.0	0.0	90.0	44.80	2,822.4	0.0	2,822.4	-32.00	-2,016.0	0.0	-2,016.0
Summation (Σ)									10,107.6	10,107.6	2,822.4		-3,288.6		-2,016.0
Balancer A			2,980.5	274,156	819.9	32.0	-164.4°	12.80	10,494.3	-10,107.6	-2,822.4	0.00	0.0	0.0	0.0
Balancer B			1,098.2	274,156	301.4	44.8	31.5°	0.00	0.0	0.0	0.0	12.80	3,657.3	3,288.6	2,016.0
SUM														0.00	0.00

Solution Summary

Plane	mr (g*mm)	θ (°)
Balancer A	2,980.5	-164.40
Balancer B	1,098.2	31.51

where:

$$\begin{aligned} \star (\text{force})A &= (((\Sigma \text{force} \cdot b \cdot \cos \theta)^2 + (\Sigma \text{force} \cdot b \cdot \sin \theta)^2)^{.5})/C \\ \star\star (\text{force})B &= (((\Sigma \text{force} \cdot a \cdot \cos \theta)^2 + (\Sigma \text{force} \cdot a \cdot \sin \theta)^2)^{.5})/C \\ \bullet \tan(\theta)A &= -(\Sigma \text{force} \cdot a \cdot \sin \theta) / -(\Sigma \text{force} \cdot a \cdot \cos \theta) \\ \bullet\bullet \tan(\theta)B &= -(\Sigma \text{force} \cdot b \cdot \sin \theta) / -(\Sigma \text{force} \cdot b \cdot \cos \theta) \end{aligned}$$

and:

$$\begin{aligned} \star (\text{mr})A &= (\text{force})A \cdot 1e6 / \omega^2 \\ \star\star (\text{mr})B &= (\text{force})B \cdot 1e6 / \omega^2 \end{aligned}$$

Free Speed, No Drag Applied Yet

mass 1		mass 2		Balancing plane		Loading	
m1 (g)	202.0	m2 (g)	75.6	A (mm)	32.0	RPM under load	10,000
r1 (mm)	7.0	r2 (mm)	7.0	B (mm)	44.8	Drag force (N)	0.0
$\phi 1 (^{\circ})$	0.0	$\phi 2 (^{\circ})$	0.0	C = B-A (mm)	12.8	angle ($^{\circ}$)	90.0
Z1 (mm)	19.4	Z2 (mm)	43.0			Placement (mm)	0.0

Plane	m (g)	r (mm)	mr (g*mm)	ω^2 (rad/s) ²	Force (N)		Z (mm)	θ	From Input			Balancing Plane A			Balancing Plane B		
					$m\omega^2$	Dmag			b	a	force "b"	force "D"cos θ	force "a"	force "D"sin θ	force "a"	force "D"cos θ	force "a"
1	202	7	1,414.0	1,098.623	1,550.8		19.4	0.0	25.40	39,385.9	39,385.9	0.0	-12.60	-19,637.9	0.0		
2	75.6	7	529.2	1,098.623	830.3		43.0	0.0	1.80	1,044.8	1,044.8	0.0	11.00	6,383.7	0.0		
Dmag						0.0	0.0	90.0	44.80	0.0	0.0	0.0	-32.00	0.0	0.0		
Summation (Σ)										40,430.5	40,430.5	0.0		-13,154.2	0.0		
Calculated Values																	
Balancer A			2890.51	1,098.623	3,278.5		32.0	-164.4	12.80	41,977.1	-40,430.5	-11,269.6	0.00	0.0	0.0		
Balancer B			2891.0962	1,098.623	1,205.4		44.8	31.5	0.00	0.0	0.0	12.80	15,428.2	13,154.2	8,064.0		
SUM										0.00	0.00	-11,269.6		0.00	8064.00		

Plane	mt (g ² mm)	θ ($^{\circ}$)
Balancer A	2,990.5	-164.4
Balancer B	1,089.2	31.5

where: (from solution when drag is applied)

★ $(mr)/A = 2,990.51 \text{ (g}^*mm)$

$\bullet (\theta)A = -164.4^\circ$

$$(mr)B = 1,099.2 \text{ (g}^* \text{mm)}$$

$$\therefore (\theta)B = 31.5^\circ$$

It will be understood that in order to facilitate comparison, masses m_1 and m_2 are shown in Fig. 3 and set forth in TABULATIONS II and III as being identical to those of Fig. 2 and TABULATION I, and that the location of the balancing masses m_A^1 and m_B^1 are disposed in the same planes in which balancing masses m_A and m_B are disposed.

The balance sketch of Fig. 3 and TABULATION II differ from Fig. 2 and TABULATION I in that they take into consideration torque applied to pad 22 in opposition to the driven rotation of assembly 20 and pad 22 about axis 18 under a predetermined working condition and the angular velocity of masses m_1 , m_2 , m_A^1 and m_B^1 , which was determined to be 5000 rpm for the sample machine under such predetermined working conditions. As a result, the sizes and angular orientations of masses m_A^1 and m_B^1 relative to axial plane 60 required to balance the sample machine under a predetermined working condition differs from the size and orientation of masses m_A and m_B previously determined to be required to balance such machine while in an unloaded condition. The drag force causing the torque under the predetermined working condition of the sample machine was determined to be 63 Newtons. The drag force lies within the previously-mentioned reference plane, that is, the surface of pad 22 disposed in abrading engagement with the work surface, and passes through the center of pad 22 tangent to the orbital path of such center about axis 18.

TABULATION III differs from TABULATION II in that drag is omitted in order to illustrate how the sample machine, once balanced by masses m_A^1 and m_B^1 sized and arranged, as shown in Fig. 3, becomes unbalanced when

subject to an unloaded rotational velocity determined to be 10,000 rpm.

The drag force acting on pad 22 under a predetermined working condition may be determined by first operating the orbital machine under load, in order to establish the amount of force required to be applied by an operator normal to the pad in order that a desired work surface finishing result is best achieved, and then measuring the rotational speed of pad 22 under such working condition. Thereafter such predetermined working condition may be repeated, for instance, by employing a pad subject to noticeable deflection under a given amount of operator applied force, and by using a feedback of the vibration level characteristic of a balanced machine under the predetermined working condition to train an operator to apply a relatively constant normal force to the pad.

The measured rotational speed is then used to read the torque corresponding to such speed from a torque vs. speed curve for the sample machine. The torque read from the torque vs. speed curve is then divided by the radial distance between axes 18 and 24 to obtain a value for drag force. Having both the value of the drag force and the previously measured angular velocity, the size and locations of balancing masses m_A^1 and m_B^1 may be calculated.

It will be noted that the resultant positions of balancing masses m_A^1 and m_B^1 are not symmetrical relative to plane 60, as best shown in Fig. 5a.

As indicated above, the working condition at which a desired surface finish is obtained will determine the manner in which the sample machine is balanced, and once

balanced, it will become unbalanced when run in an unloaded condition or when, for instance, it is used to perform a different type of abrading operation characterized for example as involving a different coefficient of friction between the pad and the work surface being abraded.

It is anticipated that an orbital machine may be designed for a drag force, which is less than that which would be anticipated during a predetermined working condition, in order to reduce the vibrational level occurring in the unloaded condition of the machine, while still substantially reducing the vibration level of the machine in loaded condition below that, which would have occurred incident to balancing thereof at unloaded condition without regard to drag. Moreover, it is anticipated that an orbital machine, such as an orbital sander capable of mounting sand paper in a range of grit sizes, may be balanced for a midpoint of a range of anticipated operating conditions in order to provide for an overall reduction in vibration throughout the range of anticipated use of such sander compared to that normally encountered by balancing same only in its unloaded condition.

What is claimed is:

1. An orbital abrading machine comprising:
a housing;
a motor supported by said housing and having an output driven for rotation about a first axis;
a pad for use in abrading a work surface;
coupling means for coupling said pad to said output for movement along an orbital path of travel about said axis; and

counterbalance means for dampening vibration of said machine while said pad is engaged with said work surface to a greater degree than when said pad is not engaged with said work surface.

2. A machine according to claim 1, wherein said pad is supported for rotation about a second axis disposed parallel to said first axis and arranged for movement along said orbital path.

3. A machine according to claim 1, wherein said coupling means includes an assembly having a head portion driven for rotation about said first axis and defining a mounting recess having a second axis; bearing means supported within said mounting recess; and connecting means for connecting said pad to said bearing means for rotation about said second axis; and said counterbalance means is operable to substantially counterbalance said pad, said bearing means, said connecting means and any portion of said head portion not concentric to said first axis and substantially counterbalance a drag force acting on said pad when engaged with said work surface.

4. A machine according to claim 3, wherein said

counterbalance means includes at least first and second masses carried by said head portion to project radially of said ~~second~~ axis.

first *on 9/11/25*

5. A machine according to claim 4, wherein said first and second axes lie within a plane and said first and second masses are not bisected by said plane.

6. A random orbital abrading machine comprising:

a manually manipulatable housing;

a motor supported by said housing and having an output driven for rotation about a first axis;

a pad for use in abrading a work surface; and

an assembly for coupling said output to said pad for application to said work surface and mounting said pad for free rotational movement about a second axis adapted to orbit about said first axis, and said assembly includes counterbalance means for substantially counterbalancing said pad and portions of said assembly not disposed concentric to said first axis and for substantially counterbalancing a drag force acting on said pad only when engaged with said work surface.

7. A random orbital buffing machine comprising:

a housing;

a motor supported by said housing and coupled to a drive shaft driven for rotation about a first axis;

a pad for use in buffing a work surface; and

an assembly for coupling said drive shaft to said pad for application to said work surface and mounted for free rotational movement about a second axis adapted to orbit about said first axis, and said assembly includes

means for dampening vibration due to a drag force acting on said pad when engaged with said work surface.

8. An assembly for connecting an abrasive pad to drive means of a random orbital abrading machine having a first axis of rotation, whereby to support said pad for free rotational movement about a second axis of rotation disposed parallel to said first axis, as said second axis is caused to orbit about said first axis, said assembly comprising:

a head portion adapted for connection with said drive means for rotation therewith about said first axis and defining a mounting recess;

bearing means supported within said mounting recess and defining said second axis;

means for connecting said pad to said bearing means for rotation about said second axis; and

counterbalance means for at least substantially counterbalancing said pad and portions of said assembly not disposed concentrically of said first axis and for at least substantially counterbalancing forces to which said pad is exposed during use as a result of engaging a work surface.

9. An assembly according to claim 7, wherein said first and second counterbalance means include first and second masses carried by said head portion to project in generally opposite directions radially of said second axis, and said first and second masses being arranged such that they are not bisected by said plane.

10. A portable orbital abrading machine comprising:

a manually manipulatable housing;

a motor supported by said housing;

an orbital assembly coupled to said motor for driven rotation about a first axis of rotation and adapted to support a work piece engaging abrasive pad for orbital movement about first axis; and

counterbalance means for counterbalancing at least a substantial portion of a drag force acting on said pad as a result of abrading engagement thereof with said work surface during use of said machine under work conditions producing a predetermined torque opposing driven rotation of said assembly about said axis.

11. A machine according to claim 10, wherein said counterbalance means is carried by said assembly.

12. A machine according to claim 11, wherein said assembly is directly coupled to said motor by a drive shaft of said motor.

13. In a portable random orbital abrasive machine having a motor supported within a manually-manipulated housing and coupled to an assembly supported for rotation about a first rotational axis and adapted to support a work surface engaging abrasive pad for free rotation about a second rotational axis which is arranged parallel to said first axis and orbits thereabout, the improvement comprising in combination:

means for counterbalancing at least a substantial portion of a weight of said pad and portions of said assembly not disposed concentric to said first rotational shaft and for counterbalancing at least a substantial portion of a drag force acting on said pad as a result of abrading engagement thereof with said work surface during

use of said machine under work conditions producing a predetermined torque opposing driven rotation of said assembly about said first axis.

14. The improvement according to Claim 13, wherein said first and second axes lie within a plane, and said counterbalance means includes at least first and second masses arranged to project radially of said second axis and arranged such that they are not bisected by said plane.

15. A machine according to claim 13, wherein said first and second axes lie within a plane, and said counterbalance means includes at least two masses arranged non-symmetrical relative to said plane.

16. In a portable orbital abrasive machine having a motor supported by a manually-manipulated housing and coupled to an assembly supported for rotation about an axis and adapted to support a work surface engaging abrasive pad for orbital movement about said axis, the improvement comprising in combination:

means for counterbalancing at least a substantial portion of a drag force acting on said pad as a result of abrading engagement thereof with said work surface during use of said machine under work conditions producing a predetermined torque opposing driven orbital movement of said pad.

ABSTRACT OF THE DISCLOSURE

A random orbital abrading machine having a housing, a drive shaft driven by a housing mounted motor for rotation about a first axis of rotation, an assembly for connecting a work surface abrading pad or the like to the drive shaft, wherein the pad is adapted to undergo free rotational movement about a second axis disposed parallel to the first axis of rotation, as such pad is caused to orbit about such first axis of rotation, characterized in that the assembly is designed for dampening vibration due to a drag force acting on the pad when engaged with the work surface under predetermined working conditions.

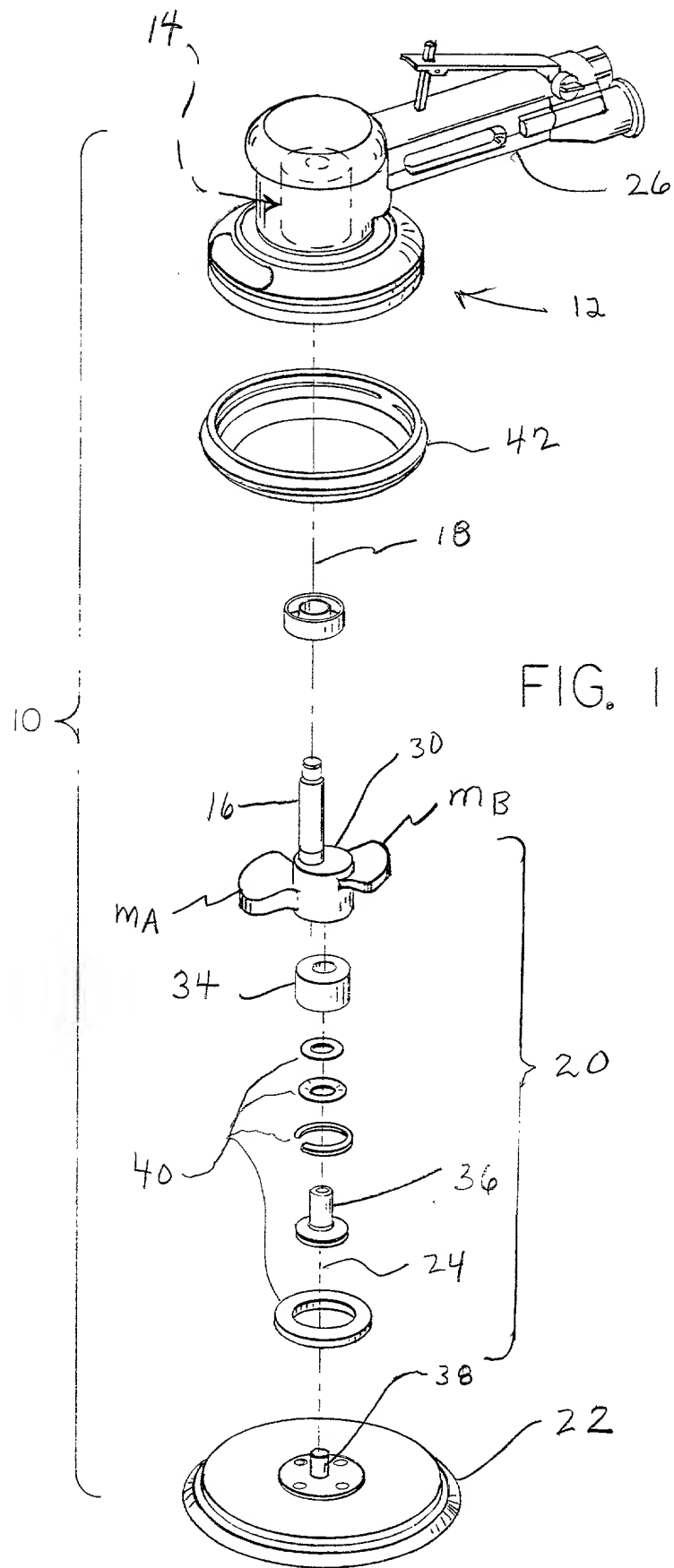
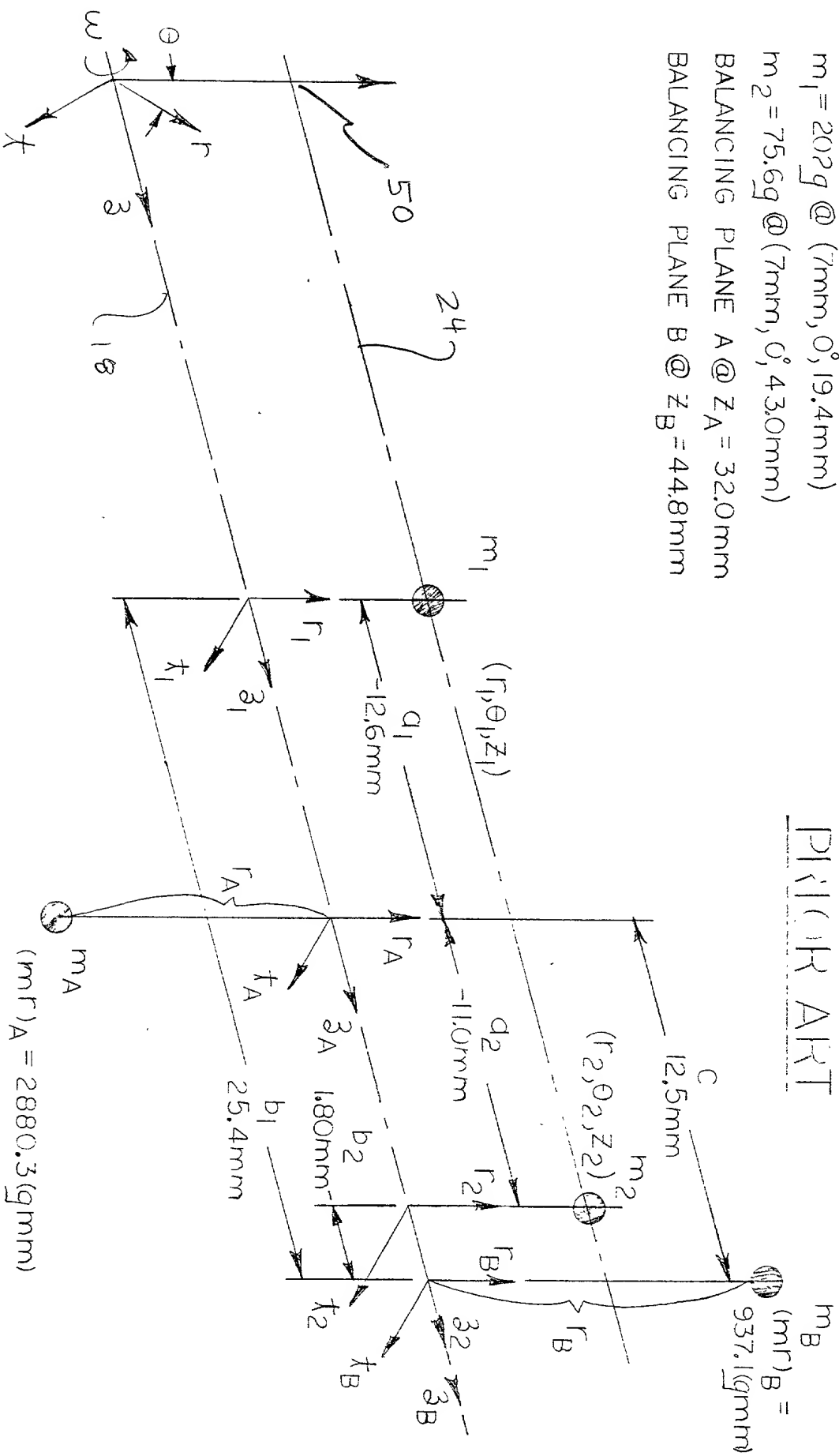


FIG. 2

PICKART

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$m_1 = 202g @ (7mm, 0^\circ, 19.4mm)$
 $m_2 = 75.6g @ (7mm, 0^\circ, 43.0mm)$
 BALANCING PLANE A @ $Z_A = 32.0mm$
 BALANCING PLANE B @ $Z_B = 44.8mm$
 LOAD = $63N @ (7mm, 90^\circ, 0mm)$
 @ $5,000rpm$



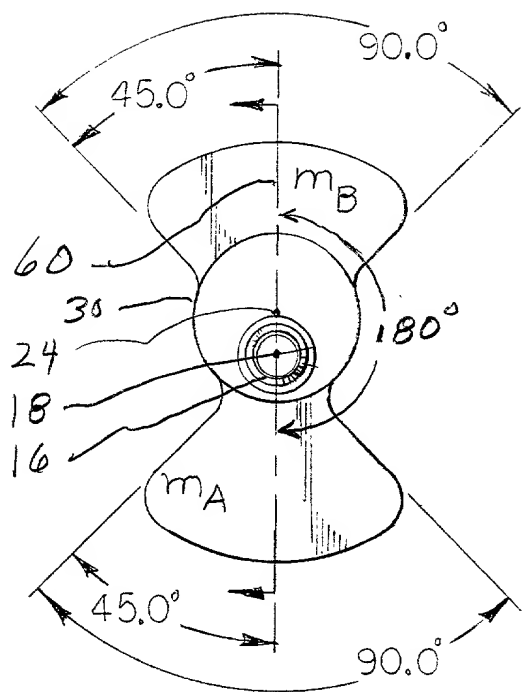


FIG. 4a

PRIOR ART

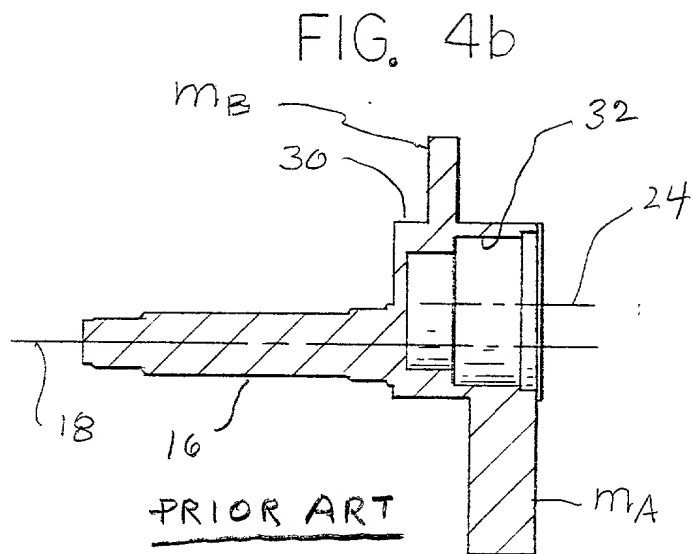


FIG. 4b

PRIOR ART

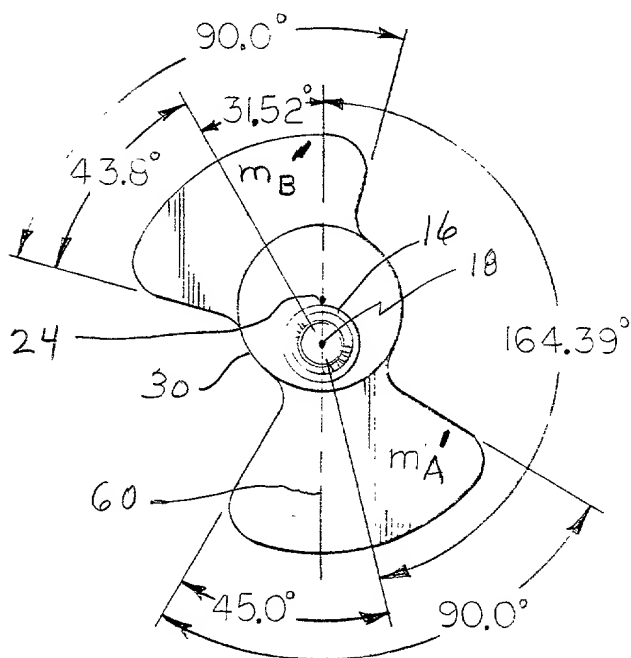


FIG. 5a

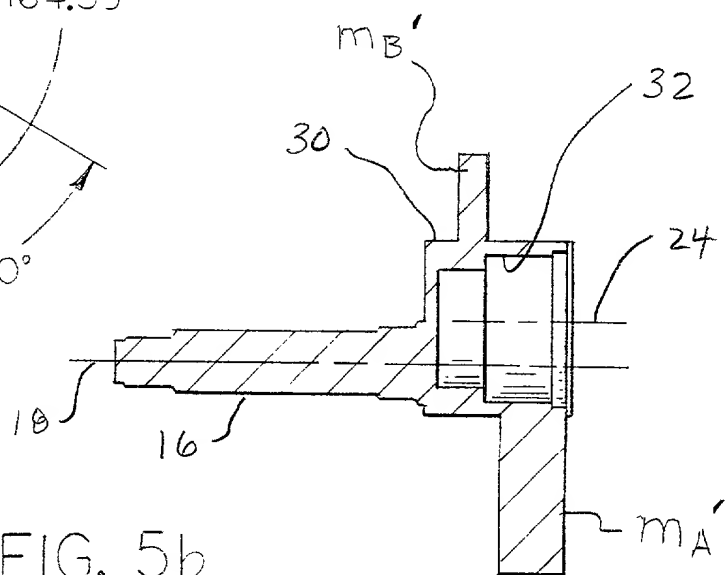


FIG. 5b

**VERIFIED STATEMENT (DECLARATION) CLAIMING SMALL ENTITY
STATUS (37 CFR 1.9(f) AND 1.27 (c)) - SMALL BUSINESS CONCERN**

Docket No.
240.988218

Serial No.

n/a

Filing Date

n/a

Patent No.

n/a

Issue Date

n/a

Applicant/ **Frank D. Lehman**
Patentee:

Invention: **BALANCER FOR ORBITAL ABRADING MACHINE**

I hereby declare that I am:

- ☐ the owner of the small business concern identified below:
☒ an official of the small business concern empowered to act on behalf of the concern identified below:

NAME OF CONCERN: **Dynabrade, Inc.**ADDRESS OF CONCERN: **8989 Sheridan Drive, Clarence Center, New York 14031-1490**

I hereby declare that the above-identified small business concern qualifies as a small business concern as defined in 13 CFR 121.3-18, and reproduced in 37 CFR 1.9(d), for purposes of paying reduced fees under Section 41(a) and (b) of Title 35, United States Code, in that the number of employees of the concern, including those of its affiliates, does not exceed 500 persons. For purposes of this statement, (1) the number of employees of the business concern is the average over the previous fiscal year of the concern of the persons employed on a full-time, part-time or temporary basis during each of the pay periods of the fiscal year, and (2) concerns are affiliates of each other when either, directly or indirectly, one concern controls or has the power to control the other, or a third party or parties controls or has the power to control both.

I hereby declare that rights under contract or law have been conveyed to and remain with the small business concern identified above with regard to the above identified invention described in:

- ☒ the specification filed herewith with title as listed above.
☐ the application identified above.
☐ the patent identified above.

If the rights held by the above-identified small business concern are not exclusive, each individual, concern or organization having rights to the invention is listed on the next page and no rights to the invention are held by any person, other than the inventor, who could not qualify as an independent inventor under 37 CFR 1.9(c) or by any concern which would not qualify as a small business concern under 37 CFR 1.9(d) or a nonprofit organization under 37 CFR 1.9(e).

Each person, concern or organization to which I have assigned, granted, conveyed, or licensed or am under an obligation under contract or law to assign, grant, convey, or license any rights in the invention is listed below:

- ☒ no such person, concern or organization exists.
☐ each such person, concern or organization is listed below.

FULL NAME

ADDRESS

☐ Individual ☐ Small Business Concern ☐ Nonprofit Organization

FULL NAME

ADDRESS

☐ Individual ☐ Small Business Concern ☐ Nonprofit Organization

FULL NAME

ADDRESS

☐ Individual ☐ Small Business Concern ☐ Nonprofit Organization

FULL NAME

ADDRESS

☐ Individual ☐ Small Business Concern ☐ Nonprofit Organization

Separate verified statements are required from each named person, concern or organization having rights to the invention averring to their status as small entities. (37 CFR 1.27)

I acknowledge the duty to file, in this application or patent, notification of any change in status resulting in loss of entitlement to small entity status prior to paying, or at the time of paying, the earliest of the issue fee or any maintenance fee due after the date on which status as a small entity is no longer appropriate. (37 CFR 1.28(b))

I hereby declare that all statements made herein of my own knowledge are true and that all statements made on information and belief are believed to be true; and further that these statements were made with the knowledge that willful false statements and the like so made are punishable by fine or imprisonment, or both, under Section 1001 of Title 18 of the United States Code, and that such willful false statements may jeopardize the validity of the application, any patent issuing thereon, or any patent to which this verified statement is directed.

NAME OF PERSON SIGNING:

TITLE OF PERSON SIGNING

OTHER THAN OWNER:

ADDRESS OF PERSON SIGNING:

9715 The Maples
Clarence, NY 14031

SIGNATURE:

DATE: January 25, 1999

Docket No.
240.988218

Declaration and Power of Attorney For Patent Application

English Language Declaration

As a below named inventor, I hereby declare that:

My residence, post office address and citizenship are as stated below next to my name,

I believe I am the original, first and sole inventor (if only one name is listed below) or an original, first and joint inventor (if plural names are listed below) of the subject matter which is claimed and for which a patent is sought on the invention entitled

BALANCER FOR ORBITAL ABRADING MACHINE

the specification of which

(check one)

☒ is attached hereto.

☐ was filed on _____ as United States Application No. or PCT International Application Number _____ and was amended on _____

(if applicable)

I hereby state that I have reviewed and understand the contents of the above identified specification, including the claims, as amended by any amendment referred to above.

I acknowledge the duty to disclose to the United States Patent and Trademark Office all information known to me to be material to patentability as defined in Title 37, Code of Federal Regulations, Section 1.56.

I hereby claim foreign priority benefits under Title 35, United States Code, Section 119(a)-(d) or Section 365(b) of any foreign application(s) for patent or inventor's certificate, or Section 365(a) of any PCT International application which designated at least one country other than the United States, listed below and have also identified below, by checking the box, any foreign application for patent or inventor's certificate or PCT International application having a filing date before that of the application on which priority is claimed.

Prior Foreign Application(s)

Priority Not Claimed

_____ (Number)	_____ (Country)	_____ (Day/Month/Year Filed)	<input type="checkbox"/>
_____ (Number)	_____ (Country)	_____ (Day/Month/Year Filed)	<input type="checkbox"/>
_____ (Number)	_____ (Country)	_____ (Day/Month/Year Filed)	<input type="checkbox"/>

I hereby claim the benefit under 35 U.S.C. Section 119(e) of any United States provisional

(Application Serial No.)

(Filing Date)

(Application Serial No.)

(Filing Date)

(Application Serial No.)

(Filing Date)

I hereby claim the benefit under 35 U. S. C. Section 120 of any United States application(s), or Section 365(c) of any PCT International application designating the United States, listed below and, insofar as the subject matter of each of the claims of this application is not disclosed in the prior United States or PCT International application in the manner provided by the first paragraph of 35 U.S.C. Section 112, I acknowledge the duty to disclose to the United States Patent and Trademark Office all information known to me to be material to patentability as defined in Title 37, C. F. R., Section 1.56 which became available between the filing date of the prior application and the national or PCT International filing date of this application:

(Application Serial No.)

(Filing Date)

(Status)
(patented, pending, abandoned)

(Application Serial No.)

(Filing Date)

(Status)
(patented, pending, abandoned)

(Application Serial No.)

(Filing Date)

(Status)
(patented, pending, abandoned)

I hereby declare that all statements made herein of my own knowledge are true and that all statements made on information and belief are believed to be true; and further that these statements were made with the knowledge that willful false statements and the like so made are punishable by fine or imprisonment, or both, under Section 1001 of Title 18 of the United States Code and that such willful false statements may jeopardize the validity of the application or any patent issued thereon.

POWER OF ATTORNEY: As a named inventor, I hereby appoint the following attorney(s) and/or agent(s) to prosecute this application and transact all business in the Patent and Trademark Office connected therewith. *(list name and registration number)*

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Second inventor's signature	Date
Residence	
Citizenship	
Post Office Address	